D. REAR SPAR SHEARWEB FORWARD LAMINATION

STEP D-1

8-7 4,0

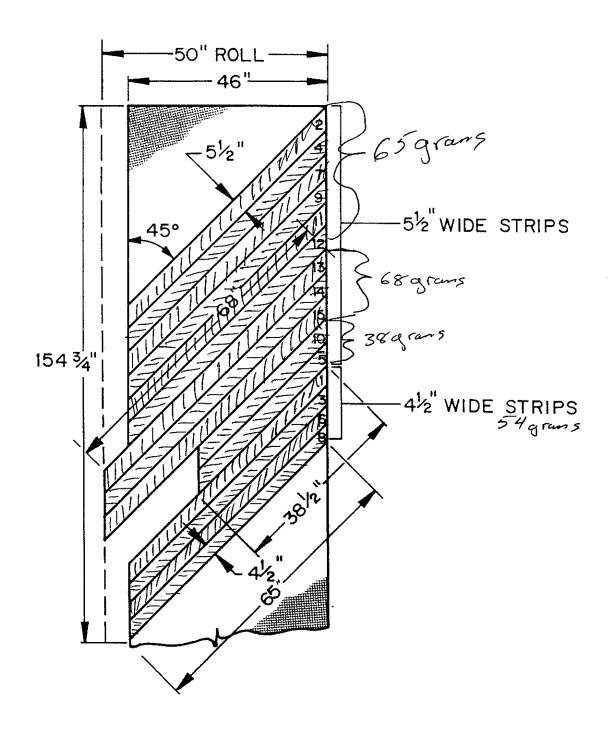


FIGURE (C-13)

STODDARD-HAMILTON

MODEL ASSEMBLY NAME REVISION DATE VOLUME PAGE
GLASAIR II RG WING ASSEMBLY
I C-19

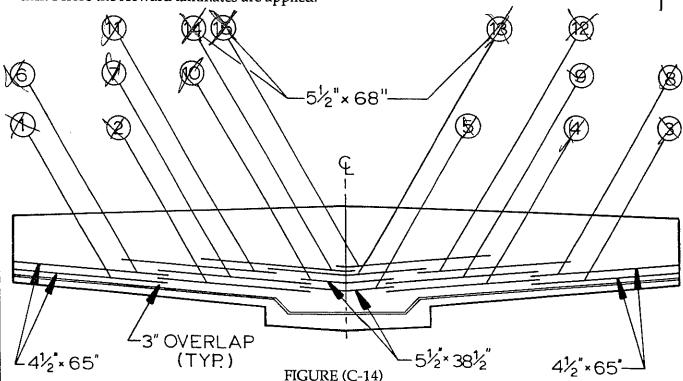
From 50" wide cloth, cut strips on the 45° bias to the following dimensions, as shown in FIGURE (C-13).

- (6) 5-1/2" x 65"
- (4) 4-1/2" x 65"
- (2) 5-1/2" x 38-1/2"
- (3) 5-1/2 X 68"

<u>NOTE</u>: Roll the strips up individually and place a rubber band around each roll. The rolls should then be appropriately numbered with a fine tip felt marker to prevent mixups during lamination.

STEP D-2

<u>NOTE</u>: At this time, the builder may wish to look ahead to subsection E, on pages C-22 through C-24, and accomplish an initial fit-up of the rear spar shear web. The shear web will be easier to trim before the forward laminates are applied.



First, wet the area to be laminated on the forward side of the shear web. Laminate in the order 1,2,3,4 etc. as shown in FIGURE (C-14). Use standard laminating techniques. Overlap the laminates 3" as shown in FIGURE (C-15). The rear spar web laminates should overlap the lower wing spar cap approximately 1". Refer to FIGURE (C-16) for a cross section of the laminate at the tip.

8-8 complete up to #11 6.0 8-16 completed ALL 1.0

STODDARD-HAMILTON
AIRCRAFT INCORPODATED

				THE PARTY OF THE P		
ı	MODEL	ASSEMBLY NAME	REVISION	DATE	VOLUME	PAGE
I	GLASAIR II	RG WING ASSEMBLY	F	10/01/91	I	C-20
Į			<u> </u>		1	

NOTE: Laminates 1, 3, 6, 8, use the 4-1/2" x 65" strips. Laminates 5 and 10 use the 5-1/2" x 38-1/2" strips. Laminates 2, 4, 7, 9, 11, and 12 use the 5-1/2" x 65" strips. Laminates 13, 14, and 15 use the 5-1/2" x 68" strips.

Let the laminates cure.

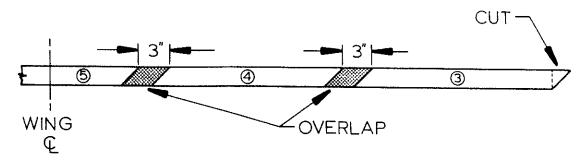


FIGURE (C-15)

COMPLETED []

STODDARD-HAMILTON AIRCRAFT, INCORPORATED

MODEL ASSEMBLY NAME REVISION DATE VOLUME PAGE C-21